

Amendments To the Claims:

Please amend the claims as shown.

1. (currently amended) ~~Stationary blade-A turbine vane (12) for a turbine (1), in particular for a gas turbine (1) for generating electrical energy, with comprising:~~

a hollow sectional element (22) which extends radially with respect to the rotor (3) and which has a transverse platform (23) at each of its ends, whereby the sectional element (22) is surrounded by hot working medium (11); ~~with~~

a hollow inset (20), located in the sectional element (22), which stretches between the two platforms (23), ~~is having~~ a certain distance from the inside (28) of the sectional element (22) and ~~has having~~ a base (35) which faces one of the two transverse platforms (23); ~~with the~~

~~a coolant (K)~~ flowing in radially through the other platform (23) into the hollow space (21) of the inset (20) and at least partially flowing out through baffle cooling openings (29) provided on the inset (20) aligned to the inside (28); and

a recess (24) that is provided in the platform (23) located immediately opposite the base (35), ~~characterized in that wherein~~

the inset (20) stretches into the recess (24) so that areas with reduced predefined flow rates are present for forming a particle trap in the base area (30) of the inset (20).

2. (currently amended) ~~Stationary blade-A turbine vane (12) according to Claim 1,~~
~~characterized in that wherein~~ the base (35) has at least one outlet opening (31) for the coolant (K) to produce a defined pressure gradient in the base area (30).

3. (currently amended) ~~Stationary blade-A turbine vane (12) according to Claim 1 or 2,~~
~~characterized in that wherein~~ the inset (20) in the base area (30) is set at a distance from the recess (24) so that appropriate outflow cross-sections (S1, S2, S3) are available for the coolant (K).

4. (currently amended) ~~Stationary blade-A turbine vane (12) according to Claim 1, 2 or 3,~~
~~characterized in that wherein~~ the recess (24) is formed as a platform penetration (39) which can be sealed from the outside by means of a cover plate (32).

5. (currently amended) ~~Stationary blade~~ A turbine vane (12) according to Claim 4,
~~characterized in that wherein~~

the cover plate (32) is welded to the platform (23) from the outside.

6. (currently amended) ~~Stationary blade~~ A turbine vane (12) according to Claims 2 to 5,
~~characterized in that wherein~~ the outlet opening (31) is a drill hole.

7. (currently amended) ~~Stationary blade~~ A turbine vane (12) according to Claim 6,
~~characterized in that wherein~~ the outlet opening (31) has a larger hole diameter than the baffle
cooling openings (29).

8. (currently amended) ~~Stationary blade~~ A turbine vane (12) according to Claim 6,
~~characterized in that wherein~~ the hole diameter of the outlet opening (31) is between 1 mm and
3 mm.

9. (currently amended) ~~A~~ Tturbine (8) ~~with a stationary blade (12)~~ according to one of the
~~preceding claims comprising:~~

a compressor section for compressing air;

a combustion section for receiving the air and a fuel and combusting the fuel/air
mixture to generate a working gas;

a turbine vane adapted for receiving the working gas comprising:

a hollow sectional element which extends radially with respect to the rotor
and which has a transverse platform at each of its ends, whereby the sectional element is
surrounded by hot working medium;

a hollow inset, located in the sectional element, which stretches between
the two platforms, having a certain distance from the inside of the sectional element and having
a base which faces one of the two transverse platforms;

a coolant flowing in radially through the other platform into the hollow
space of the inset and at least partially flowing out through baffle cooling openings provided on
the inset aligned to the inside; and

a recess that is provided in the platform located immediately opposite the base, wherein the inset stretches into the recess so that areas with reduced predefined flow rates are present for forming a particle trap in the base area of the inset.

10. (new) A Turbine vane for a turbine according to Claim 1, wherein the turbine is a gas turbine.

11. (new) A turbine vane according to Claim 2, wherein the inset in the base area is set at a distance from the recess so that appropriate outflow cross-sections are available for the coolant.

12. (new) A turbine vane according to Claim 2, wherein the recess is formed as a platform penetration which can be sealed from the outside by means of a cover plate.

13. (new) A turbine vane according to Claim 3, wherein the recess is formed as a platform penetration which can be sealed from the outside by means of a cover plate.

14. (new) A turbine vane according to Claim 3, wherein the outlet opening is a drill hole.

15. (new) A turbine vane according to Claim 4, wherein the outlet opening is a drill hole.

16. (new) A turbine vane according to Claim 5, wherein the outlet opening is a drill hole.

17. (new) A turbine with a turbine vane according to Claim 9, wherein the base has at least one outlet opening for the coolant to produce a defined pressure gradient in the base area.

18. (new) A turbine with a turbine vane according to Claim 9, wherein the inset in the base area is set at a distance from the recess so that appropriate outflow cross-sections are available for the coolant.

19. (new) A turbine with a turbine vane according to Claim 9, wherein the recess is formed as a platform penetration which can be sealed from the outside by means of a cover plate.